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Comparing Cyclic Tension-Compression Effects on CoCrFeMnNi High-Entropy Alloy and Ni-Based Superalloy

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Abstract: An equal-molar CoCrFeMnNi, face-centered-cubic (fcc) high-entropy alloy (HEA) and a nickel-based superalloy are studied using in situ neutron diffraction experiments. With continuous measurements, the evolution of diffraction peaks is collected for microscopic lattice strain analyses. Cyclic hardening and softening are found in both metallic systems. However, as obtained from the diffraction-peak-width evolution, the underneath deformation mechanisms are quite different. The CoCrFeMnNi HEA exhibits distinct lattice strain and microstructure responses under tension-compression cyclic loadings.

Keywords: in-situ neutron diffraction; low-cycle fatigue; high entropy alloy; Ni-based superalloy

1. Introduction

Fracture toughness and fatigue behavior are the key to the practical applications of high entropy alloys (HEAs) [1]. Hemphill et al.'s [2] four-point-bending fatigue experiments suggest that the fatigue behavior of Al_{0.5}CoCrCuFeNi HEA is affected by microstructural defects of oxides and microcracks while Tang et al.'s [3] high-cycle four-point-bending fatigue experiments reveal that nanotwinning is the major deformation behavior before fracture in a Al_{0.5}CoCrCuFeNi two-phase HEA. Despite the convenience of four-point-bending high-cycle fatigue (HCF), more detailed fatigue mechanisms, especially in the low-cycle fatigue (LCF) in HEAs, need to be further explored, since the deformation behaviors under HCF and LCF regimes are very different, as suggested by Chen et al.'s review [4].

Thurston et al. [5] report that fatigue behavior of the Cantor alloy is comparable to the austenitic stainless steels and twinning induced plasticity (TWIP) steels. Recently, Niendorf et al.'s [6] tension-compression LCF experiments indicate the influence of planarity of slip and partial reversibility of deformation on the unobvious cyclic hardening at given plastic strain in Fe₅₀Mn₃₀Co₁₀Cr₁₀ HEA. There are few reports revealing the polymorphism of CoCrFeMnNi, which is induced by tension [7] and compression [8–10], respectively. Moreover, the cyclic tension-compression effects as well as predominant mechanisms during the LCF process in CoCrFeMnNi still remain elusive. Meanwhile, the cyclic-loading deformation in face-centered-cubic (fcc) nickel-based superalloy has been extensively examined in which lattice strain asymmetry associated with dislocation density evolution governs the irreversible plastic deformation [11–14]. We herein investigate the in-situ

tension-compression cyclic loading effects and the dominant deformation behaviors of the CoCrFeMnNi HEA in comparison with the commercially popular nickel-based superalloy by exploiting in-situ neutron diffraction measurements.

2. Materials and Methods

The nickel-based superalloy HASTELLOY C22™ (Ni-21Cr-17Mo in weight percentage) was annealed to dissolve most precipitates into the fcc matrix structure. C22™ has been chosen as a reference because this Ni-based superalloy has been extensively studied using neutron diffraction and other complimentary experiments [11–17]. Several correlations from diffraction-data characteristics, microscopic images and underneath deformation mechanisms have been found as benchmarks and references. Hence, we follow the same protocol and experimental design performed on the C22™ to apply to the current CoCrFeMnNi HEA examinations. In addition, the LCF experiments and cyclic-loading deformation mechanisms of the C22™ are clarified and cited in many articles [11–16].

For the CoCrFeMnNi HEA, the sample was fabricated by vacuum arc-melting (arc) with equal molar compositions of constituent powders whose purity is more than 99.9% (in weight percent) and the diffraction characteristics of the CoCrFeMnNi HEA are also well studied for its microstructure [10,18–20].

The selected gauge volume for both samples was approximately 120 mm³ to associate the microstructure with the bulk properties. During cyclic deformation of the C22™ alloy, in situ neutron diffraction measurements with the Spallation Neutron Source were carried out at the Spectrometer for Materials Research at Temperature and Stress (SMARTS) beamline of Los Alamos Neutron Science Center (LANSCE). The in situ neutron diffraction measurements for LCF of the HEA were conducted at TAKUMI beamline in Materials and Life Science Experimental Facilities at J-PARC, Japan. The loading direction is 45° relative to the incident neutron beam direction, with two detector banks located at ±90° concomitantly recording diffraction vector patterns in the loading and transverse directions. Both the in situ neutron diffraction LCF tests were conducted at room temperature with the maximum tensile strain of 1% and the maximum compressive strain of −1%. The cyclic frequency was 1 Hz. The neutron diffraction profiles were analyzed by applying pseudo-Voigt function for single peak fitting with the General Structure Analysis Software (GSAS, Los Alamos National Laboratory, version 1, Los Alamos County, New Mexico, United States) [21] and Convolutional Multiple Whole Profile (CMWP) [22] software, following the previously used protocol [13,15].

3. Results

3.1. Chemical Composition of Alloying Elements in CoCrFeMnNi

We employed JEOL 6500 scanning electron microscopy (SEM, Japan Electron Optics Laboratory, Tokyo, Japan) with energy-dispersive x-ray (EDX, Japan Electron Optics Laboratory, Tokyo, Japan) spectroscopy for determining the elemental composition. The EDX analysis in Figure 1 discloses the alloy composition of CoCrFeMnNi with an equal weight percentage distribution of each constituent element, determining the Co₂₀Cr₂₀Fe₂₀Mn₂₀Ni₂₀ HEA in the present study.

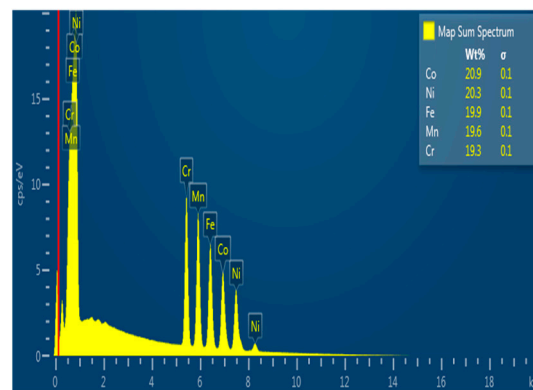


Figure 1. EDX spectrum of the CoCrFeMnNi HEA.

3.2. Low-Cycle Fatigue Experiments

Figure 2 presents the macroscopic stresses evolution versus certain selected fatigue cycles measured at the maximum strain of 1% and the minimum strain of -1% for C22TM and CoCrFeMnNi alloys. Both the C22TM and HEA performed similar cyclic hardening followed by softening tendency, however, the microscopic mechanisms underlying their cyclic hardening/softening-structural transformation were in different behaviors. Three distinct regions of the fatigue cycles based on the applied stress development can be clearly observed. During continuous tension-compression LCF deformation, the C22TM alloy revealed the cyclic hardening within the 100th cycle region with increasing applied stress (stage I), followed by the cyclic softening upon the 1000th cycle with a gradual decrease in applied stress (stage II) and remaining saturated (stage III) until the 2500th cycle. Compared to the C22TM, the HEA experienced the hardening/softening transition at the 10th fatigue cycle, and then cyclic softening. At the same strain of tensile (1%) or compressive (-1%) deformation, an increase in stress up to the hardening/softening transition on the C22TM was more pronounced than that on the HEA, which demonstrated the greater hardening behavior of the C22TM alloy.

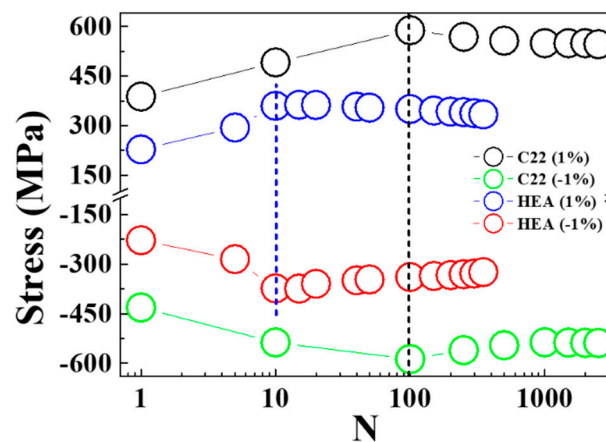


Figure 2. The stress evolution at the maximum tension (1%) and compression (-1%) as a function of fatigue cycle (presented in log scale) for both the C22TM and CoCrFeMnNi alloys.

3.3. Lattice Strain Evolution during Tension-Compression Cyclic Loadings

The lattice strain evolution was calculated based on the variations in the d-spacing of the refined neutron diffraction peak positions of lattice planes as a function of fatigue cycle with respect to the initial d-spacing before deformation. Figure 3 depicts the lattice strain-stress dependence of the {311} plane at the maximum tensile (1%) and compressive (-1%) strains in both transverse and axial loading directions since {311} grain families are the least affected crystallographic planes by intrinsic microstructure deformation rather than external loadings in fcc polycrystalline materials

under loading [12,23]. The yellow arrows show the trends of lattice strain evolution under cyclic loadings. Within the cyclic hardening stage in the early fatigue cycles, we can observe a positive correlation between the elastic strain and applied stress in both C22™ and HEA, which seems to follow Hooke's law as Suzuki and Egami reported for cyclic deformation [24]. Specifically, C22™ has a nearly linear relationship up to the hardening/softening transition. However, although the lattice strain-stress relationship of HEA is also positive up to the 10th fatigue cycle, the trend is not linear. Several reports suggest that such a nonlinear trend may be governed by the texture, patterned dislocation structure, or other anisotropy of the materials [11].

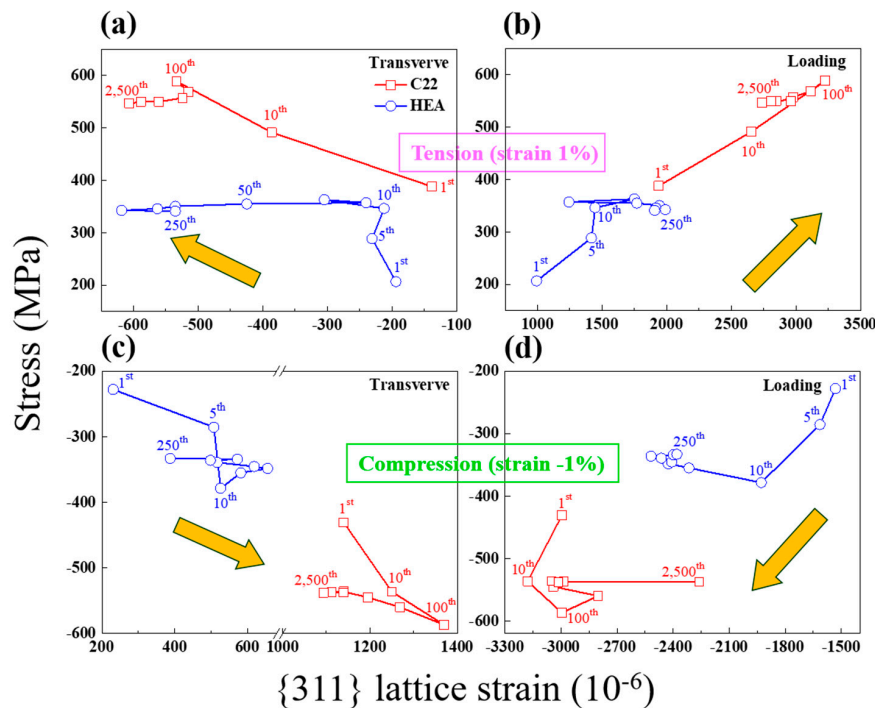


Figure 3. The {311} lattice strain-stress relationship at the maximum tensile strain of 1% in (a) transverse and (b) axial loading; and at the maximum compressive strain of −1% in (c) transverse and (d) axial loading directions for both the C22™ and CoCrFeMnNi alloys.

In the hardening cycles of stage I, both the C22™ and HEA displayed the same propensity in which the lattice strain increases with the number of fatigue cycles under both tensile and compressive deformation in both axial loading and transverse directions, ascribed to the cyclic hardening behavior in this region. However, the increase of elastic strain in the C22™ was much greater than that of the HEA under maximum tension shown in Figure 3a,b. Meanwhile, the lattice strain evolution of the C22™ and HEA was almost the same under maximum compressive deformation within the cycling hardening region. In stage II, there was an opposite trend of lattice strain evolution between the two alloys. The lattice strain of HEA continued to increase while that of C22™ experienced an initially slight decrease within the cyclic softening region. Furthermore, the elastic strain of C22™ increases under tension in Figure 3a but decreases under compression in Figure 3c from the 250th to the 1000th fatigue cycle for the grains oriented along the transverse direction. A similar tendency was also obtained for the grains of C22™ oriented along the axial loading direction in tension in Figure 3b and compression in Figure 3d. The deviation of elastic strain within the cyclic softening region was ascribed to the development of the intergranular strains [16]. In stage III, there was a gradually decrease of lattice strain of C22™ under both tensile and compressive tests for grains oriented along both axial loading and transverse directions after the 1000th cycle. Similar to the C22™ alloy, the lattice strain of HEA also depicted a slight decrease after the 200th cycle. The major discrepancy in the lattice strain evolution of

the C22TM and HEA was further clarified in terms of the dominant development of dislocation slip activities and other fatigue-induced defects such as stacking faults and twins.

4. Discussion

In this work, we determine the evolution of average dislocation density (n) and dislocation-wall spacings (D) by using CMWP software [22] in fitting the neutron diffraction profiles. Figure 4a depicts the dislocation density and dislocation wall spacings as a function of fatigue cycle at the maximum strain of 1% in both C22TM [13] and CoCrFeMnNi alloys along the axial loading direction. For CoCrFeMnNi, the dislocation density sharply increased together with an increasing number of fatigue cycles in the very early stages in the axial loading direction and reached the maximum value at the 40th cycle, then decreased with further fatigue cycles undergoing the softening cycles. The decrease in the density of the randomly distributed dislocations after the hardening cycles seen in the CoCrFeMnNi was analogous to that obtained in the C22TM alloy [13]. The average distances between the dislocation-wall structures are largest at a very early cycle because the annealed C22TM alloy has fewer cell structures than Mughrabi's dislocation wall [25] in the beginning [17]. The C22TM alloy mainly has randomly distributed dislocations before cyclic loading [17]. Upon cyclic loading, the average distances between the dislocation-wall structures dropped within the hardening region, and remained constant with higher fatigue cycles undergoing the softening and saturation regimes. Due to dislocation movement and interaction, a more orderly arrangement of dislocations associated with the formation of the Mughrabi's dislocation-wall for cell structures resulted in a decrease and final saturation of the dislocation-wall distance when the dislocation densities approached a more stable state in CoCrFeMnNi. The dislocation hardening was demonstrated to be the main deformation mechanism of the cyclic hardening effect in the C22TM alloy. The calculated dislocation-wall-spacing evolution within the hardening region of HEA is similar to that of C22TM but the dislocation density evolution of HEA is quite different from that of C22TM, implying that there are other microstructure formations coupled with cyclic deformation activities in governing the LCF regime. It has been reported that mechanical nanotwinning is the major deformation mechanism before fatigue crack initiation and propagation in HEA [3,4,26,27].

The generation of different kinds of defects associated with the dislocation density evolution under LCF was further elucidated. As reported by Ungar, each kind of defect is simultaneously governed by distinct diffraction peak aberrations such as peak shift, peak broadening, peak asymmetry, anisotropic peak broadening, and peak shape [28]. We first examined the deformation fault probability of CoCrFeMnNi obtained directly from the changes in only diffraction peak separations, which is applied only for fcc structure, as summarized by Warren [29]. Using this method, we can predict the presence of deformation fault probability based on only peak-shift correlations between $(1\ 1\ 1)$ and $(2\ 0\ 0)$ of polycrystalline metals in fcc structure, specifically, the displacement of d_{111} - d_{200} separation toward smaller or larger deviation between these two groups of lattice planes, as shown in Figure 4b. However, we cannot distinguish whether the deformation fault is intrinsic or extrinsic stacking faults. To attain the quantitative probability of intrinsic, extrinsic stacking faults, and twinning, a more complete and reliable analysis of CMWP program [22] was further employed to resolve the convoluted diffraction data and the corresponding results are presented in Figure 4c.

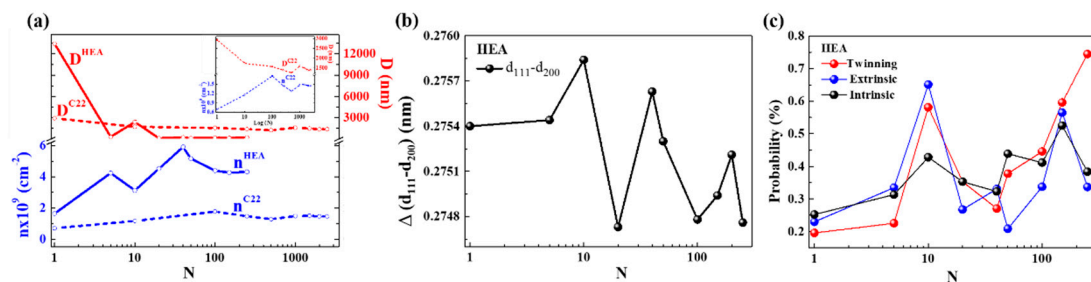


Figure 4. (a) The dislocation density and dislocation-wall-spacing evolution in C22™ and CoCrFeMnNi alloys. The magnification of n and D of C22™ alloy is indicated in the inset. (b) The difference in d -spacing between diffraction peaks 111 and 200 of CoCrFeMnNi. (c) Twining, extrinsic and intrinsic stacking faults defects as a function of fatigue cycle (presented in log scale) in CoCrFeMnNi.

It can be seen from CMWP fitting results in Figure 4c that the formation probability of twinning, extrinsic and intrinsic stacking faults defects started to slightly increase with the dislocation density evolution and the extrinsic stacking fault plays a major role in deformation in the hardening cycles. More twinning and extrinsic stacking faults were generated easily, and they governed the cyclic hardening behavior. Further increase of dislocation densities led to the decrease of twinning, and the probability of extrinsic and intrinsic stacking faults in the first half of softening cycles. When the dislocation densities were in the more stable condition, there was a slight variation of extrinsic and intrinsic stacking faults but an obvious increasing trend of twinning probability, suggesting the dominant role of twinning in the latter half of softening cycles. The formation of nanotwins and stacking faults is attributed to the low stacking fault energy (SFE) of HEA.

5. Conclusions

During low-cycle fatigue experiments, both the CoCrFeMnNi and C22™ alloys display obvious cyclic hardening and softening regimes, however, the CoCrFeMnNi indicated fewer hardening cycles comparing to the C22™ alloy. The lattice strain-stress relationship of these two alloys are quite different, indicating that dominant deformation-induced microstructures were distinct under tension-compression cyclic loadings. The twinning, extrinsic and intrinsic stacking faults are found to be formed during cyclic hardening regimes, and twinning is the most predominant defect in governing the main deformation behavior of CoCrFeMnNi in the latter half of softening regimes. The role of twin formation subjected to tension-compression cyclic loadings on CoCrFeMnNi is revealed, which may be useful for HEA design for fatigue endurance.

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Conflicts of Interest: The authors declare no conflicts of interest.

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